

## Computational Fluid Dynamics at Marshall Space Flight Center

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# The Role of Computational Fluid Dynamics At Marshall Space Flight Center

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Emerging Horizons of Turbomachinery Technology

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### Outline of Presentation

- MSFC's Roles and Responsibilities
- Fluid Dynamics Division
- CFD in the Design and Development Process at MSFC
- Cost vs. Value of CFD
- Accuracy
- Future Needs
- Summary

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### MSFC's Roles and Responsibility (as related to CFD)

- MSFC is the Center of Excellence for Space Propulsion Technology within NASA
- A primary MSFC thrust is to pursue activities that lower the cost of access to space
  - » Operational systems: Space Shuttle Main Engine (SSME); Solid Rocket Motors
  - » Experimental or in Development: X-33, X-34, Reusable Launch Vehicle, Rocket Based Combine Cycle Engines, etc.
- MSFC performs numerous functions related to its Center of Excellence mission
  - » Program development and management
  - » Engineering: design, analysis, coordination/integration
  - » Testing: "cold flow" testing, "hot fire" testing of components, subsystem demonstrations, materials testing, structural testing, etc.

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### Fluid Dynamics Division

- The Fluid Dynamics Division is part of the Structures and Dynamics Lab
  - » The Lab provides most of the advanced analytical support to MSFC programs
    - CFD, Stress, Structural Dynamics, Thermal, etc.
    - Lab also provides structural design and testing organizations and cold flow testing
- The Fluid Dynamics Division is divided into two branches
  - » Experimental and Analytical
- The two branches are mutually complimentary and are integrated
  - » Most test supported with analyses and most analyses verified with testing
- Experimental facilities include:
  - » Two closed-loop pump test stands (150 Hp and 350 Hp) and one open-loop, all using water as the test fluid
  - » One long-duration blowdown turbine test stand that uses air as the test fluid
  - » Tri-sonic wind tunnel
  - » Rocket engine nozzle test facility that uses
  - » Solid Rocket motor technology test rig

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### Fluid Dynamics Analysis Branch

- The analysis branch include several related disciplines
  - » CFD, acoustics (including high frequency data reduction, induced environments
- Approximately 16 full time CFD users (out of a group of 27)
- We apply rather than develop CFD
  - » Limited CFD code enhancements and customization
- We apply CFD primarily to reduce the cost of design and development
  - » Maximize analysis fidelity early while the design can be modified inexpensively
    - Do not wait until the design is set in concrete to perform sophisticated analyses
  - » Increase likelihood that the first “final design” will be the last “final design”
    - Continue supporting refinement of design throughout the detail design process
    - Assess the design against requirements, assess for undesirable flow features
    - Provide better environments for structural and thermal design and analyses
  - » Maximize testing value by assessing test plan, instrumentation location, etc.
  - » Support failure investigations

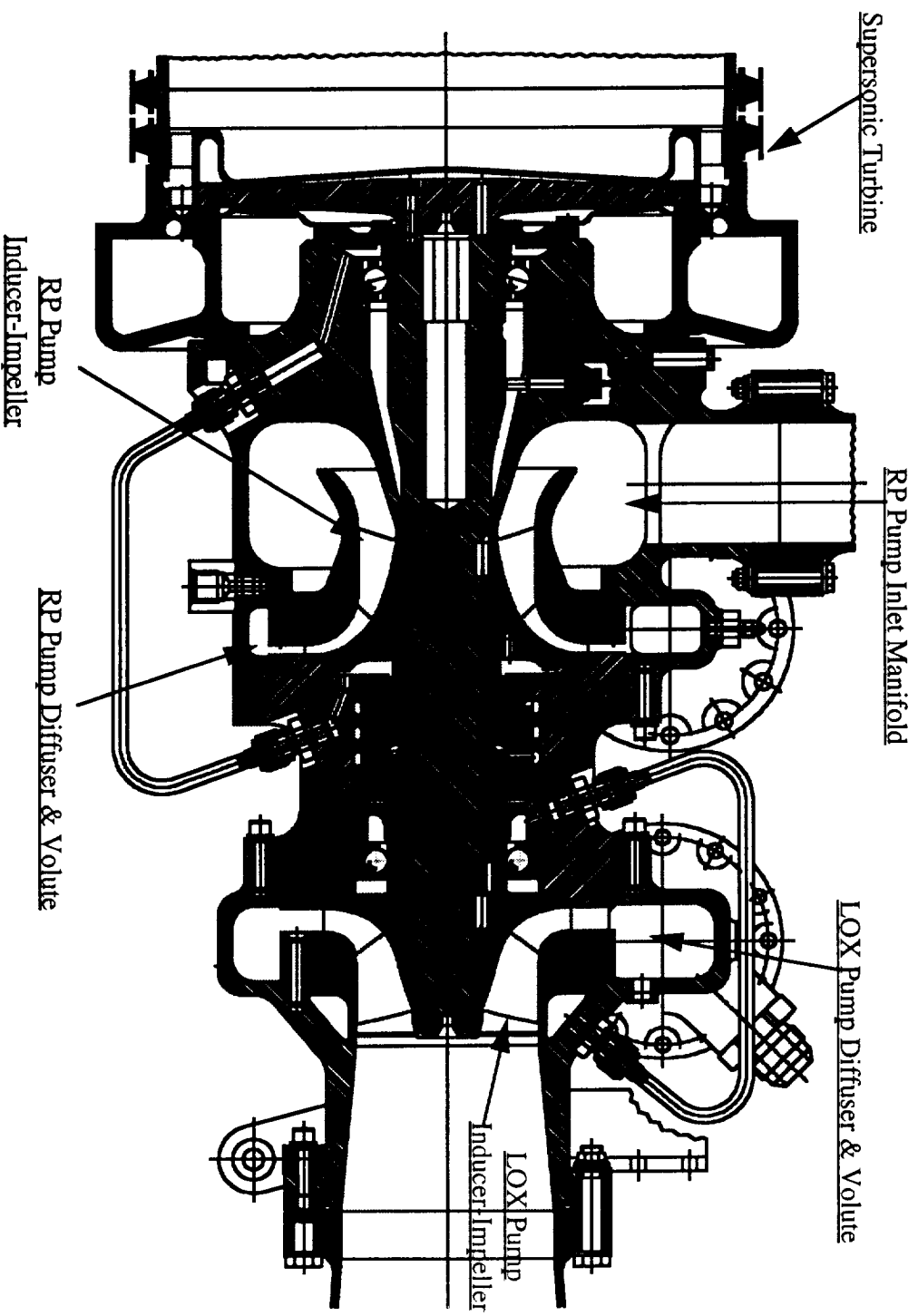
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### CFD in the Design and Development Process at MSFC

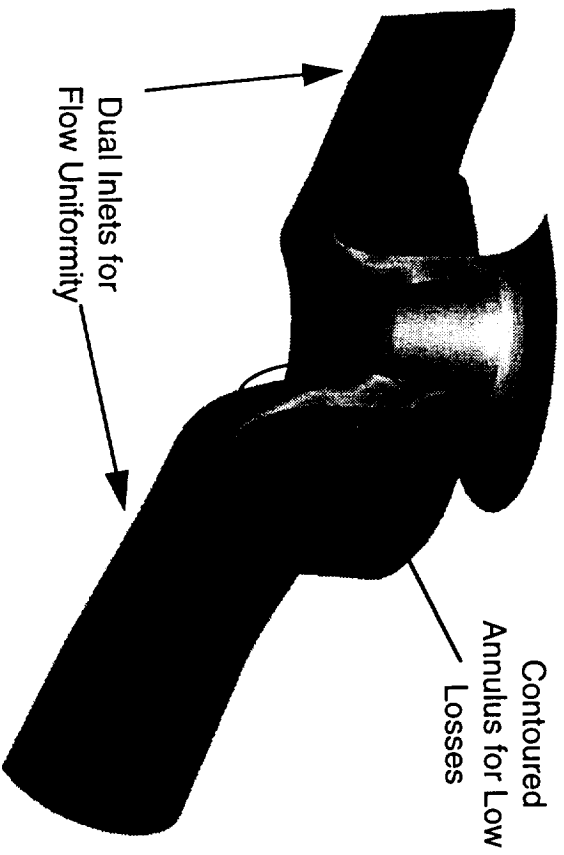
- CFD is applied to a broad range of components and subsystems
  - » Pump and turbine feedlines and feed manifolds
    - Incompressible or compressible, steady, 3D
  - » All primary and secondary flow paths in pumps
    - Incompressible, single phase, steady, 2D and 3D
  - » All primary and secondary flow paths in turbines
    - Compressible, subsonic to supersonic, steady and unsteady, 2D and 3D
  - » Combustion chambers
    - Compressible, reacting, multi-species, hydrogen and hydrocarbons, 2D and 3D
  - » Rocket engine nozzles and plumes
    - subsonic to supersonic, reacting, multiple species, internal and external flow, 2D and 3D
  - » Launch vehicles ascent
    - subsonic to supersonic, integrated engine with engine plume on, reacting, multi-species, various altitudes, 3D

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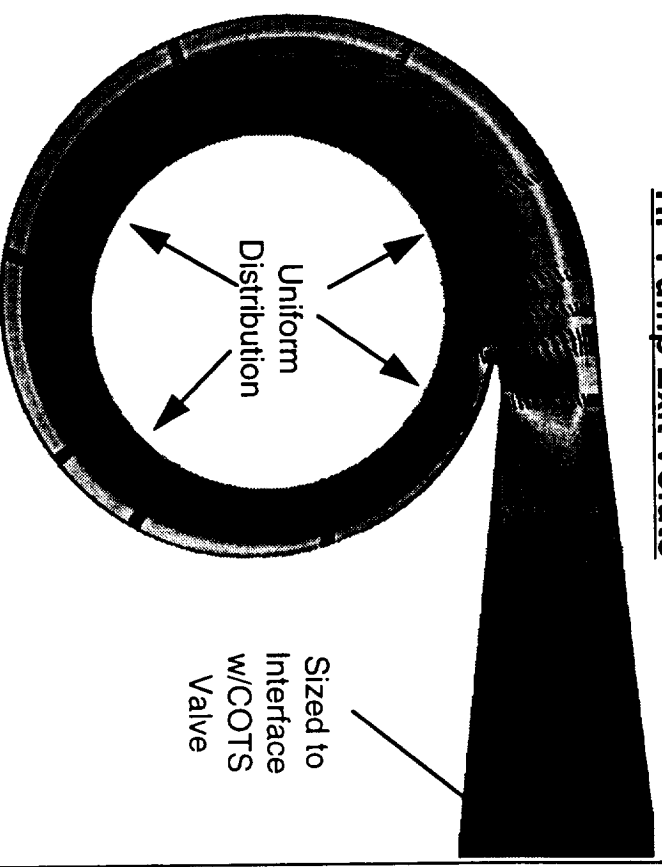


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RP Pump Inlet Manifold



RP Pump Exit Volute



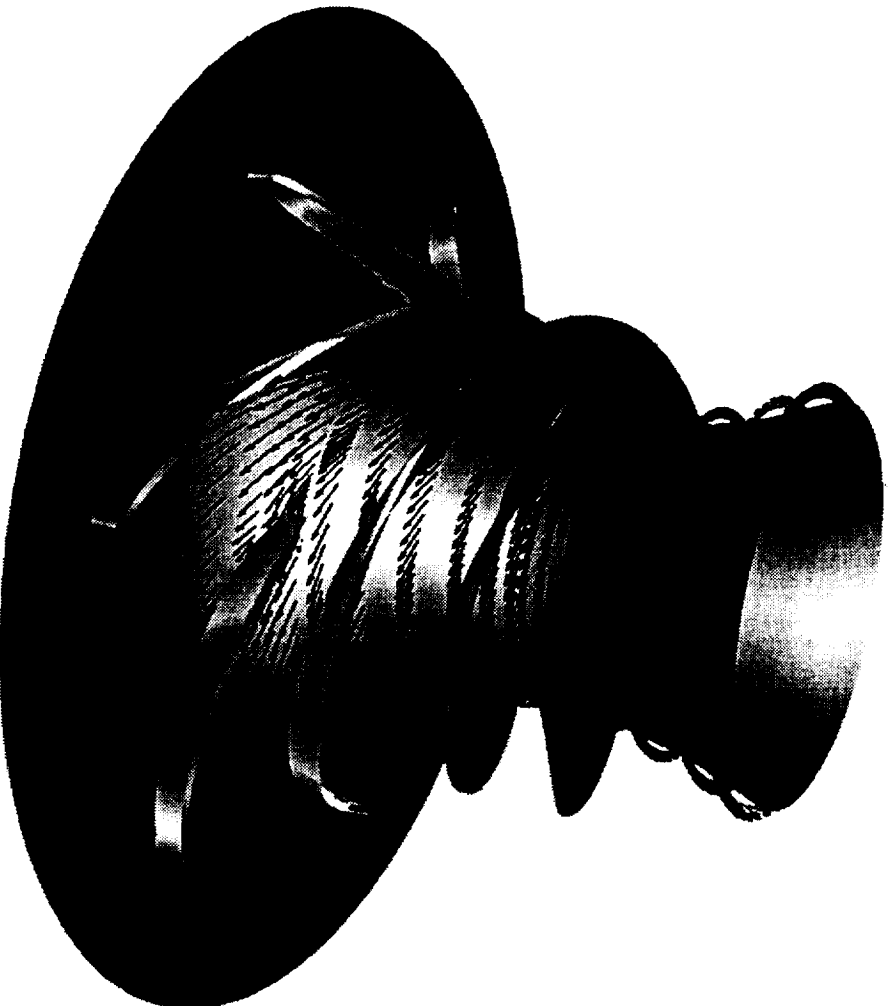


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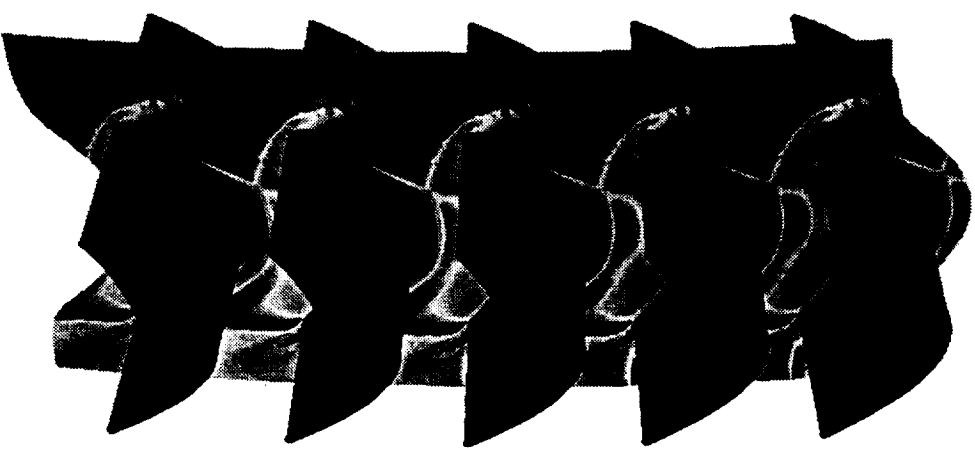
### RP Inducer-Impeller

blade pressures and velocity vectors



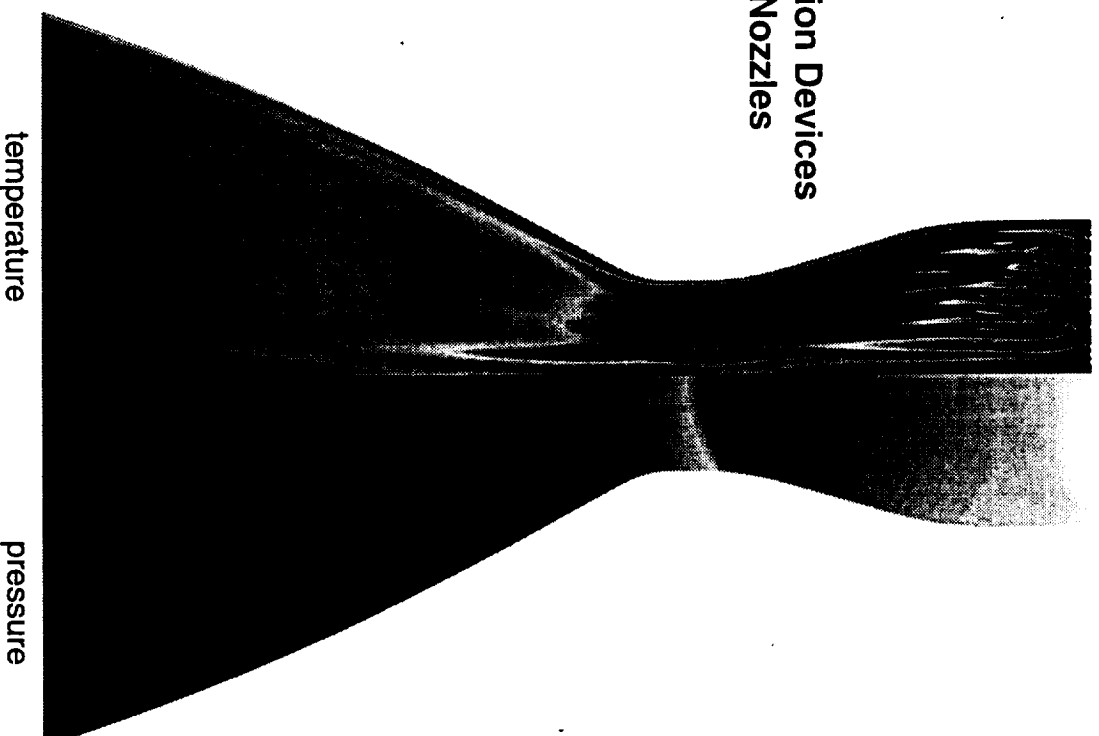
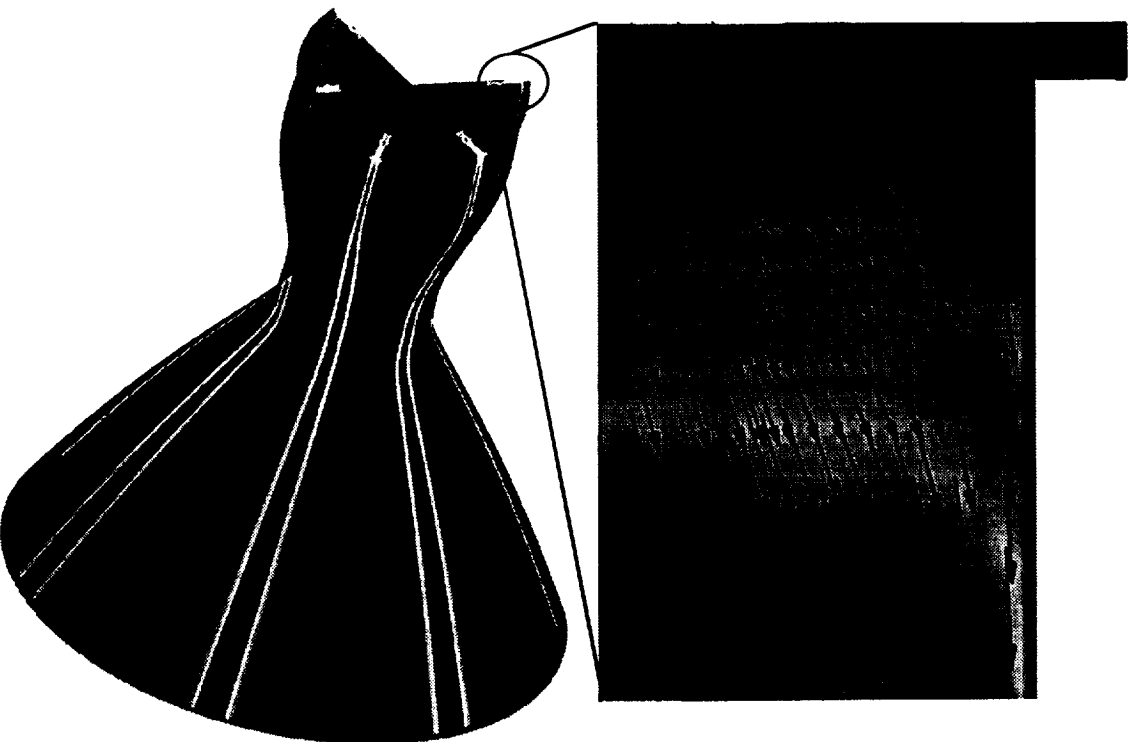
### Turbine Rotor Blades

midspan absolute mach number



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### Combustion Devices and Nozzles



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### CFD in the Design and Development Process at MSFC

- This range of applications imposes unique constraints on our software
  - » We cannot afford to have specialized codes for type of application
    - Management, training, support nightmare
    - Rely primarily on three codes (2 of which are public domain codes)
    - More efficient use of personnel
  - » We support development of the codes that we use
    - Insufficient market for much of our unique applications for commercial development
  - » We require generalize grid generation and solution visualization tools
    - Generally separate from the CFD code
  - » A tendency towards specialization among the personnel
    - Grid or visualization specialists
    - Turbomachinery vs. combustion device; internal vs. external flows
    - Familiarity with the specifics of the particular flow being analyzed is key to successful application of CFD
- Minimize leasing of software, opt for public domain or purchase
  - » Due to uncertainty of year-to-year funding (government constraint)

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### Cost vs. Value of CFD

- Rocket engine hardware is extremely expensive to build and test
  - » Weight and performance at a premium
  - » Designs generally have little margin
  - » Environments are very severe
  - » Nearly every new program outside the experience base
    - Very little direct scaling is possible
- Relatively high engineering and development cost due to limited production
- The cost of one SSME engine test is greater than the average yearly hardware and software cost for the CFD group
- CFD represents a relatively small investment and overhead relative to the impact it has on the hardware
  - » Most of the designs analyzed are impacted by the CFD results

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### Cost vs. Value of CFD (continued)

- Computing cost are continuously decreasing
  - » Code robustness and convergence properties have improved
  - » Computing cost have dramatically decreased
    - Super computer speed on \$35,000 workstations
      - Sub-net of inexpensive workstations presents tremendous potential for parallelization and surge capacity
    - Mini-super computers with multiprocessors (up to 18) for less than \$500,000
      - Compared to \$4 - \$5 million a few years ago
  - » High chip speeds and inexpensive memory cost have led to improvements in interactive grid generation and visualization tools
  - » Personnel cost are the most significant over the long term
    - Not too many good part-time CFD users; it is a full time job
    - Reasonable theoretical and/or experimental background is very important
    - Excellent computer skills including programming skills
    - Must have some level of proficiency in all steps of the analysis process
    - Must be able to relate CFD results to hardware design implications
    - Dedicated hardware and system software specialist required even for small groups

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### Accuracy

- Predictable error is more useful than inconsistent accuracy
  - » Can adjust or account for the former
  - » Often rely on relative differences in assessing design changes
- Achievable accuracy varies from application to application
  - » Higher accuracy can be expected for pressure or body force dominated flows than for viscous forces (and mixing) dominated flows
    - Radial impeller performance near the design point can be predicted very accurately
    - Diffuser performance prediction accuracy is not nearly as good (test final design)
- Lack of accuracy often caused by improper analyses application
  - » Improperly applied or located boundary conditions
  - » Steady analysis of inherently unsteady flows (i.e., diffusers)
- Benchmark only to the level required to verify code/model fundamentals
  - » Establish code suitability and grid requirements
  - » Elaborate benchmark demonstrations have limited value

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### Future Needs

- Practical incompressible time-accurate capability
  - » Impeller-vaned diffuser interaction
- Cavitation modeling capability
  - » Every single rocket engine pump cavitates during normal operation
  - » Tremendous numerical and physical modeling challenge
  - » Inherently unsteady
- Hybrid schemes for internal and external flows
  - » Facilitate grid generation of complex geometries
    - lifting body with engine, inducer with tip clearance, etc.
  - » Must address pre- and post-processing issues
- All sorts of automation:
  - » Automated grid generation for specific components
  - » Automated solver initialization based on past results
  - » Automated links between the geometric model (CAD) and CFD, and between CFD and the disciplines requiring the CFD output (stress, thermal, etc.)

## **Computational Fluid Dynamics at Marshall Space Flight Center**

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### **Summary**

- CFD is used at MSFC to support the design and development of propulsion and vehicle systems
- CFD is part of the process that includes testing
- Broad range of applications at MSFC leads to unique CFD requirements
- CFD is good value relative to the cost of testing and redesign
- Known inaccuracy is important for proper application of CFD results
- Unsteady pumps flows, cavitation, and complex geometries require CFD developer's attention